

## Saddle

Developer:	Meca Enterprises Chris Rosencutter chris@mecaconsulting.com
Description:	Perform analysis of a horizontal vessel supported on two saddles. The program performs a complete analysis of the vessels based upon L. P. Zick's analysis procedure. The program allows the user to specify the wind load, thermal loads, liquid load, and seismic loads on the vessel. A complete analysis is also performed on the saddle, calculating stresses on bolts, base plate, web, stiffeners, and wear plate.
Platform:	Saddle is a Microsoft Excel spreadsheet. It requires MS Excel 2000 or later to run.
Units:	English or Metric (Click of the button automatically toggles all inputs and results)
Sample Output:	<p>The following pages are the output from the Saddle spreadsheet. As an example, we have simulated an example of the Zick analysis performed in the "Pressure Vessel Handbook" by Megyesy. The example in Megyesy does not address the design of the saddle itself, so there is only a comparison of the vessel stresses.</p> <p>In the output you will see a note symbol that looks like this →</p> <p>If you double click this note, you will receive further explanation of the output.</p>
Purchase:	You may purchase this program at <a href="http://www.mecaconsulting.com">www.mecaconsulting.com</a> .

# Saddle v1-1

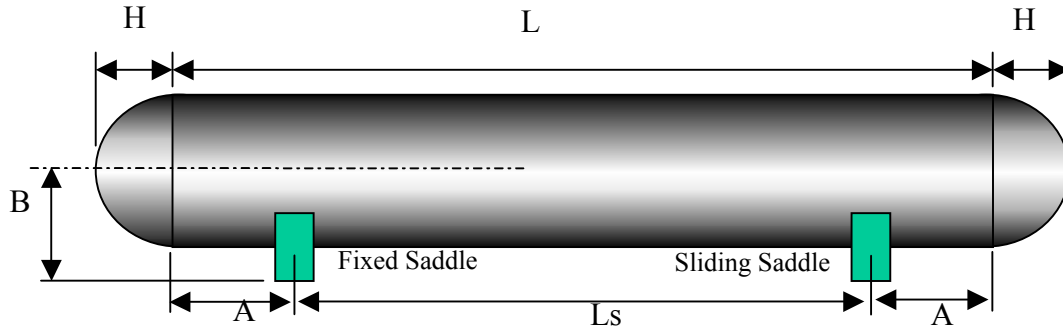
per "Pressure Vessel Design Manual" by Dennis R. Moss

Customer: Meca

Desc: Example from Pressure Vessel Handbook

Project: Pg 93

by: CR



Vessel Information		
OD	Outside Diameter of Vessel	120.0000 in
ts	Corroded Thickness of Shell	1.0000 in
P	Internal Design Pressure	250.00 psig
Pe	External Design Pressure	0.00 psig
L	Tangent to Tangent Length of Vessel	960.00 in
H	Depth of Head	21 in
A	Distance from Head Tangent to Saddle Center Line	48.00 in
B	Height from vessel centerline to bottom of saddle	69.00 in
HType	Head Type	Hemis
th	Corroded Thickness of Head	1 in
Wind Design (ASCE 7-98)		
V	Design Wind Speed	0 mph
I	Importance Factor	1
Exp	Exposure	C
Seismic Design (UBC 1997)		
Z	UBC 1997 Seismic Zone	0
I	Importance Factor	1
Sc	Soil Coefficient (SA, SB, SC, SD, or SE)	SC
Temperature		
Tinst	Installation Temperature of Vessel	70.00 Deg. F
Tmin	Minimum Temperature of Vessel	70.00 Deg. F
Tmax	Maximum Temperture of Vessel	70.00 Deg. F
u	Coefficient of Friction between Saddle and Concrete	0.45
fc	Allowable bearing pressure on concrete	750 psi
Material Properties		
S	Allowable Stress: Vessel Shell	17,500 psi
Fys	Yield Stress of Shell at Design Temperature	38,000 psi
Fy	Yield Stress of Saddle	36,000 psi
Emod	Modulus of Elasticity of Saddle	2.900E+07 psi
Fbolt	Allowable Tensile Stress on Bolts	20,000 psi
Fvbolt	Allowable Shear Stress on Bolts	10,000 psi
Av	Coefficient of Thermal Expansion of Vessel	7.00E-06 in/in/F
JE	Joint Efficiency	0.85
Weight		
W	Total Weight (If zero, program will estimate)	600,000 lbs

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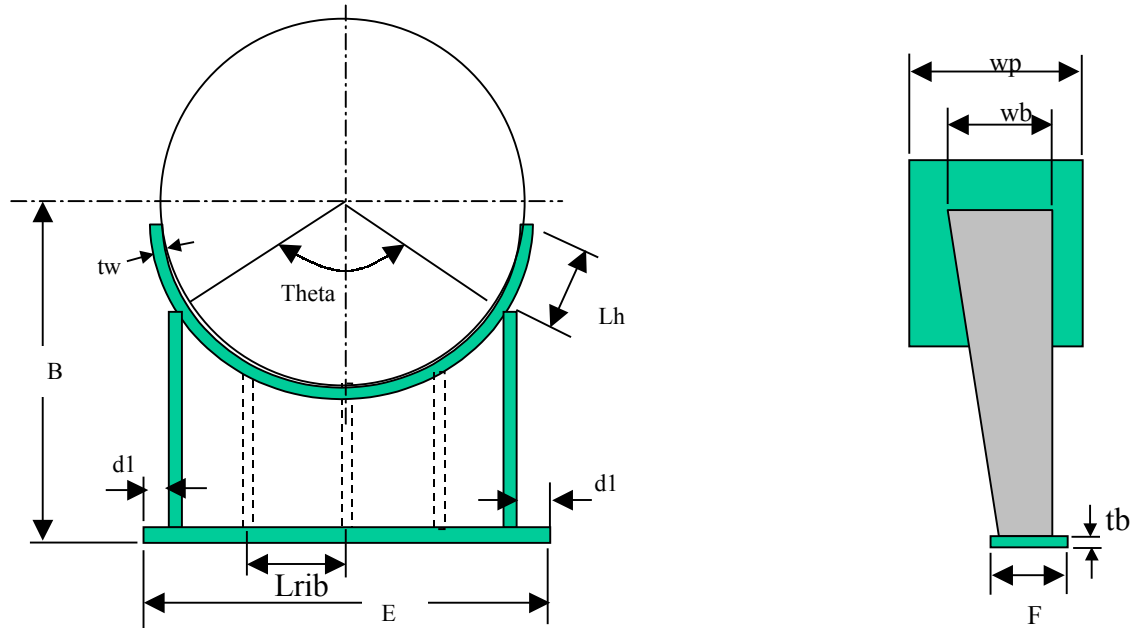
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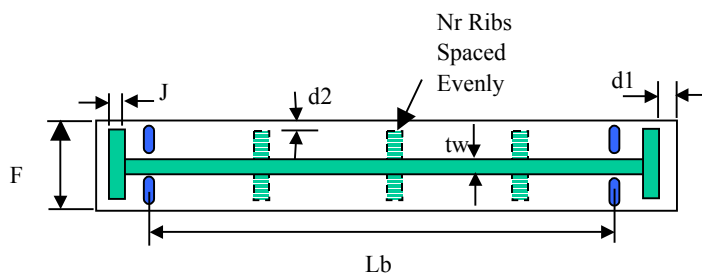
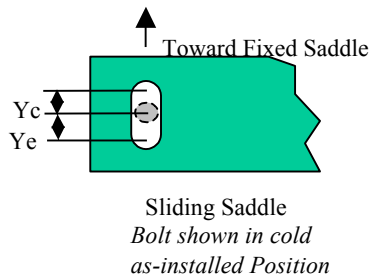
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Saddle Information (Note Click "Std Dims" button to get a standard saddle)				
E	Width of Saddle perpendicular to Longitudinal Axis	Theta =	120.15	104.000 in
F	Width of Saddle at Bottom (along Longitudinal Vessel Axis)			9.000 in
wb	Width of Saddle at Top of Saddle (along longitudinal Vessel axis)			24.000 in
d1	Distance from Outside of Baseplate to First Rib			0.000 in
tb	Thickness of Base Plate			1.000 in
wp	Width of wear plate			24.000 in
Lh	Wear Plt ext. above Horn			1.000 in
tw	Thickness of Wear Plate			0.750 in



Saddle Information				
d2	Distance from Outside of Baseplate to First Rib			0.000 in
Lb	Center to Center bolt spacing in transverse direction			80.000 in
tweb	Thickness of Web			0.75 in
J	Thickness of Ribs			0.75 in
Nr	Number of Ribs			2
Nb	Number of Anchor Bolts per Saddle			2
Dbolt	Nominal Diameter of Bolt			1.250 in
Ww	Fillet Leg Size (Web to Baseplate)			0.750 in

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Calculated Parameters		
Rm	Mean Radius of Shell	59.500 in
Rs	Radius of Shell	60.000 in
Ls	Saddle Spacing: $L-2*A$	864.000 in
Theta	Saddle Angle: $2*Atan((E-2*d1)/OD)$	120.1 Deg
ThetaW	Angle of Wear Plate	122.1 Deg
Calculate Weights		
Wtot	User Entered Total Weight	600,000 lbs
Wind Loading: (Based upon ASCE 7-98)		
De	Effective Diameter based upon Table 3-24	141.6000 in
zg	Constant from table 6-4	900.0000
Alpha	Constant from table 6-4	9.5000
Kz	$2.01*((B+OD/2)/zg)^(2/Alpha)$	1.3353
Cf	Shape Factor	0.8
Gq	Gust Factor (Rigid Structure)	0.85
qz	Wind Pressure: $0.00256*Kz*V^2*I$	0.00 psf
Afl	$PI()*(De/12)^2/4$	109 ft^2
Flw	$Af*Cf*Gq*qz$	- lbs
Aft	$De*(L+2*H)/144$	985 ft^2
Ftw	$(Aft*Cf*Gq*qz)*0.5$	- lbs
Seismic Loading: (Based upon UBC)		
Ca	Seismic Coefficient based upon Soil and Zone	0.0000
V	$2.5*Ca*I*W / R$	- lbs
Fls	$V / 1.4$	- lbs
Fts	$V / (2 * 1.4)$	- lbs
Thermal Expansion		
Ye	Maximum Expansion of Vessel	- in
Yc	Maximum Contraction of Vessel	- in
Flt	Frictional Force due to Expansion/Contraction ( $u*Wtot/2$ )	- lbs
Saddle Reactions		
Fl	Maximum Longitudinal Force: $Max(Flw, Fls, Flt, Flp)$	- lbs
Ft	Maximum Transverse Force: $Max(Ftw, Fts)$	- lbs
Qo	Operating Load on Saddles: $(Wtot)/2$	300,000 lbs
Q1	Reaction due to Long Force: $Wo/2+Fl*B/Ls$	300,000 lbs
Q2	Reaction due to Tran Force: $Wo/2+3*Ft*B/E$	300,000 lbs
Q	Maximum of Q1 or Q2	300,000 lbs

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"K" Constants from Figure 3-46									
A./Rs	=	0.8000	K5	=	0.7595				
K1	=	0.3357	K6	=	0.0369				
K2	=	1.1686	K7	=	0.6041				
K3	=	0.8775	K8	=	0.3399				
K4	=	0.4004	K9	=	0.0529				
Longitudinal Bending									
M1	$6*Q*(8*A*H+6*A^2-3*Rm^2+3*H^2)/(3*L+4*H)$					6.372E+05 in-lbs			
M2	$3*Q*((3*L^2+6*Rm^2-6*H^2-12*A*L-16*A*H)/(3*L+4*H))$					5.603E+07 in-lbs			
S1	Long. Bending @ Saddles w/o Stiffeners - Tension: $M1/(K1*Rm^2*ts)$					536 psi			
S2	Long. Bending @ Saddle w/o Stiffeners - Compression: $-M1/(K7*Rm^2*ts)$					-298 psi			
S4	Long. Bending @ Midspan: $\pm M2/(pi()*Rm^2*ts)$					5,038 psi			
Tangential Shear									
S6	Tang. Shear - shell not stiffened $A>0.5R$ : $(K2*Q/(Rm*ts))*((L-2*A)/(L+4*H/3))$					5,153 psi			
Circumferential Bending									
Check # 1	Lh $\geq$ Rm/10:		1.00	$\leq$	5.95	FALSE			
Check # 2	Is A $\leq$ 0.5*Rm:		48.00	$\geq$	29.75	FALSE			
Check # 3	Wp $\geq$ wb+1.56*(Rm*ts)^0.5		24.00	$\leq$	36.03	FALSE			
tes	Check # 1 & # 2 did not both pass: ts					1.0000 in			
tes1	Check # 3 did not both pass: ts					1.0000 in			
tes2	Check # 1 & # 2 did not both pass: ts^2					1.0000 in			
S9	Bend @ horn $L\geq 8R$ : $-Q/(4*ts*(wb+1.56*(Rm*ts)^0.5))-3*KK6*Q/(2*tes2)$					-18,694 psi			
S10	Bend @ horn $L<8R$ : $-Q/(4*ts*(wb+1.56*(Rs*ts)^0.5))-12*KK6*Q*Rs/(L*ts^2)$					-10,385 psi			
S11	Since $A>0.5*Rs \rightarrow$ Add'l Tension in Head = 0					0 psi			
S12	Circ Compression: $-KK5*Q/(tes1*(wb+1.56*(Rm*tes1)^0.5))$					-6,323 psi			
Pressure Stresses									
fx	Longitudinal Pressure Stress		$P*Rm/(2*ts)$			7,438 psi			
fp	Circumferential Pressure Stress		$P*Rm/ts$			14,875 psi			
Saddle Design - Web									
Element	b	h	Area	I	d	A*d	A*d^2	Itot	
	in	in	in^2	in^4	in	in^3	in^4	in^4	
Shell	48.167	1.000	48.2	4.01.E+00	0.50	2.408E+01	1.204E+01	1.606E+01	
Wear Plt	36.084	0.750	27.1	1.27.E+00	1.38	3.721E+01	5.117E+01	5.243E+01	
Web	0.750	37.250	27.9	3.23.E+03	20.38	5.692E+02	1.160E+04	1.483E+04	
Baseplt	9.000	1.000	9.0	7.50.E-01	39.50	3.555E+02	1.404E+04	1.404E+04	
	Area =		112.2		A*Y =		9.860E+02	I =	2.894E+04
c1	Dist from Id of Shell to Center of Gravity for Saddle ( $A*Y/Area$ )							8.79 in	
c2	Dist from Center of Gravity to Base Plate							0.21 in	
Is	Moment of Inertia of Saddle							2.894E+04 in^4	
As	Area of Saddle							112.2 in^2	
Beta	$Pi() - Theta/2$							2.093 rads	
K1	$(1+COS(Beta)-0.5*(SIN(Beta))^2)/(PI()-Beta+(SIN(Beta))*(COS(Beta)))$							0.2038	
fh	Saddle Splitting Force: $K1*Q$							61,137 lbs	
ft	Tensile Stress in Saddle: $fh/As$							545 psi	
d	$B-Rs*Sin(Theta)/Theta$							49 in	
M	$fh * d$							3.005E+06 in-lbs	
fb	Bending Stress in Saddle: $M*C1/I$							913 psi	
Saddle Design - Wear Plate									
fb	Bending Stress in Wear Plate: $6*Q*K5*wb/(8*tw^2*Rs)$							17,624 psi	
Saddle Design - Baseplate									
Ab	Bearing Area: $E * F$							936 in^2	
Bp	Bearing Pressure: $Q/Ab$							321 psi	
fbplt	$(3*Q*F)/(4*E*tb^2)$							19,471 psi	

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per "Pressure Vessel Design Manual" by Dennis R. Moss

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Saddle Design - Anchor Bolts				
Longitudinal Load				
Tension?	Is $Q_0 > Q_1$ ?	300,000	<	300,000
				FALSE
Pb	Since $Q_0 > Q_1$ then Tensile Load will exist Each Bolt: $((Q_1 - Q_0) / (N_b))$		-	lbs
Abolt	$(\pi / 4) * D_{bolt}^2$		1.23	in <sup>2</sup>
fal	Bolt Tensile Stress: $P_b / A_{bolt}$		-	psi
Shear Load (Assume Fixed Saddle takes entire load)				
fv	Shear Stress: $F_l / A_{bolt}$		-	psi
Transverse Load				
M	Transverse Moment: $F_t * B$		-	in-lbs
e	$M / Q_0$		-	in
E/6	$E / 6$		17.333	in
Since $e < E / 6 \rightarrow$ There is No Uplift				

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Saddle Design - Ribs			
Outside Ribs			
Lrib	Rib Spacing: $(E-2*d1)/(Nr-1)$	104.0000	in
Lotrib	Tributary Length: $\text{Min}(e, 0.5*Lrib)$	52.0000	in
Pr	Axial Load on Ribs: $Bp * F * Lotrib$	150,000	lbs
Ar	Area of Web and Rib: $J*(F-2*d2-tweb) + tweb*(Lotrib-d1)$	45.19	in^2
fa	Compressive Stress: $Pr / Ar$	3,320	psi
I1	$(J/12) * ((wb + F) / 2)^3$	2.808E+02	in^4
C1	$(wb + F) / 4$	8.250	in
r	Radius of Gyration: $(I1 / Ar)^{0.5}$	2.493	in
L1	Height of Saddle: $B - Rs * \text{Cos}(\text{Theta}/2)$	39.067	in
Lr	Slenderness Ratio: $L1 / r$	15.7	
Cc	$(2*PI()^2*E\text{mod}/Fy)^{0.5}$	126.1	
Fa	$(1-Lr^2/(2*Cc^2))*Fy/(5/3+3*Lr/(8*Cc)-Lr^3/(8*Cc^3))$	20,853	psi
fu	Unit Force: $FI / (2 * E)$	-	lb/ft
M	Bending Moment: $0.5 * fu * e * L1$	0.000E+00	in-lbs
fb	Bending Stress: $M * C1 / I1$	-	psi
SR	Stress Ratio: $fa/Fa+fb/Fb$	0.16	
Inside Ribs			
Litrib	Tributary Length: $\text{Min}(e, Lrib)$	104.000	in
Pr	Axial Load on Ribs: $Bp * F * Litrib$	3.000E+05	lbs
Ar	Area of Web and Rib: $J*(F-2*d2-tweb) + tweb*Litrib$	84.19	in^2
fa	Compressive Stress: $Pr / Ar$	3,563	psi
I2	$(J/12) * ((wb + F) / 2)^3$	8.640E+02	in^4
C2	$0.5 * Wb$	12.000	in
r	Radius of Gyration: $(I2 / Ar)^{0.5}$	3.204	in
L2	Height of Saddle: $B - (Rs^2-(E/2-d1-Lrib)^2)^{0.5}$	39.1	in
Lr	Slenderness Ratio: $L1 / r$	12.2	
Cc	$(2*PI()^2*E\text{mod}/Fy)^{0.5}$	126.1	
Fa	$(1-Lr^2/(2*Cc^2))*Fy/(5/3+3*Lr/(8*Cc)-Lr^3/(8*Cc^3))$	21,043	psi
fu	Unit Force: $FI / (2 * E)$	-	lb/ft
M	Bending Moment: $0.5 * fu * e * L2$	0.000E+00	in-lbs
fb	Bending Stress: $M * C2 / I2$	-	psi
SR	Stress Ratio: $fa/Fa+fb/Fb$	0.17	

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Stress Summary					
Description	Equation	SR	Result	Actual	Allowable
Tangential Shear					
Shell not stiffened	S6	0.37	PASS	5,153	14,000 psi
Circumferential Bending					
Horn of Saddle - Shell Not Stiffened	S9	0.71	PASS	(18,694)	26,250 psi
Horn of Saddle - Shell not Stiffened	S10	0.40	PASS	(10,385)	26,250 psi
Circumferential Compressive Stress	S12	0.33	PASS	(6,323)	19,000 psi
Combined Stress - Tension					
Longitudinal Tension at Saddles	S1+fx	0.54	PASS	7,974	14,875 psi
Longitudinal Bending @ Midspan	S4+fx	0.84	PASS	12,475	14,875 psi
Tension in Head	S11+fp	0.80	PASS	14,875	18,594 psi
Saddle Web					
Tensile Stress in Web	ft	0.03	PASS	545	21,600 psi
Bending Stress in Saddle	fb	0.04	PASS	913	23,760 psi
Saddle Wear Plate					
Bending Stress in Wear Plate	fb	0.74	PASS	17,624	23,760 psi
Saddle Base Plate					
Bending stress in Baseplate	fbplt	0.82	PASS	19,471	23,760 psi
Bearing pressure on Concrete	Bp	0.43	PASS	321	750 psi
Saddle - Outside Ribs					
Bending Stress	fb	0.00	PASS	-	23,760 psi
Axial Stress	fa	0.16	PASS	3,320	20,853 psi
S.R. for Bending + Axial	ftot	0.16	PASS	0.16	1.00
Saddle - Inside Ribs					
Bending Stress	fb	0.00	PASS	-	23,760 psi
Axial Stress	fa	0.17	PASS	3,563	21,043 psi
S.R. for Bending + Axial	ftot	0.17	PASS	0.17	1.00
Saddle - Inside Ribs					
Tensile Stress	fa	0.00	PASS	-	20,000 psi
Shear Stress	fv	0.00	PASS	-	10,000 psi